

# REPORT DOCUMENTATION PAGE

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MEMORANDUM FOR PR (In-House Contractor/In-House Publication)  
FROM: PROI (TI) (STINFO)

29 February 2000

SUBJECT: Authorization for Release of Technical Information, Control Number: **AFRL-PR-ED-TP-2000-039**  
Chehroudi, B. (ERC), Badakshan, A., Cohn, R., Talley, D., "Injection of Cryogenic Jets into Subcritical and Supercritical Environments"

**4<sup>th</sup> International Symposium on Liquid Space Propulsion** (Statement A)  
**Lampoldshausen, Germany, 13-15 Mar 2000 (Absolute Deadline: 09 Mar 2000)**

1. This request has been reviewed by the Foreign Disclosure Office for: a.) appropriateness of distribution statement, b.) military/national critical technology, c.) export controls or distribution restrictions, d.) appropriateness for release to a foreign nation, and e.) technical sensitivity and/or economic sensitivity.

Comments: \_\_\_\_\_  
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Signature \_\_\_\_\_ Date \_\_\_\_\_

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3. This request has been reviewed by the STINFO for: a.) changes if approved as amended, b.) appropriateness of distribution statement, c.) military/national critical technology, d.) economic sensitivity, e.) parallel review completed if required, and f.) format and completion of meeting clearance form if required

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4. This request has been reviewed by PR for: a.) technical accuracy, b.) appropriateness for audience, c.) appropriateness of distribution statement, d.) technical sensitivity and economic sensitivity, e.) military/national critical technology, and f.) data rights and patentability

Comments: \_\_\_\_\_  
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APPROVED/APPROVED AS AMENDED/DISAPPROVED

ROBERT C. CORLEY  
Senior Scientist (Propulsion)  
Propulsion Directorate

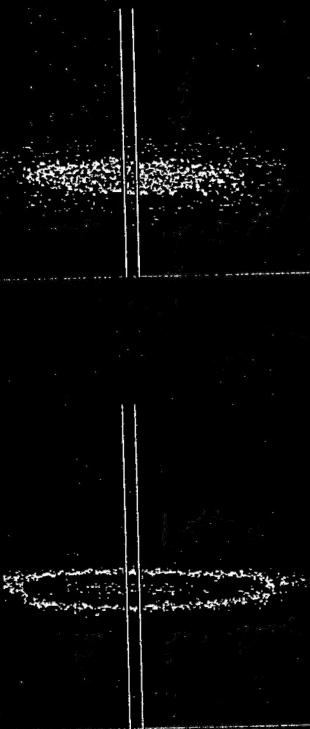
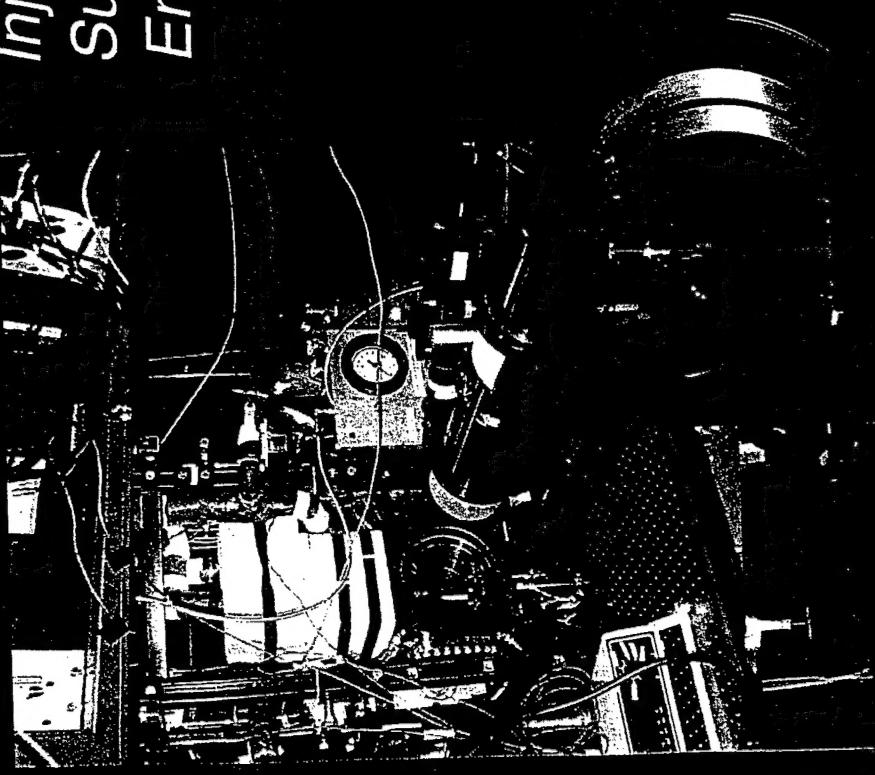
(Date)

Fourth International Symposium on Liquid Space Propulsion  
DLR - Lampoldshausen, Germany  
March 13 - 15, 2000



*Injection of Cryogenic Jets into  
Subcritical and Supercritical  
Environments:*

B. Chehroudi, A. Badakshan,  
R. Cohn, and D. Talley



Subcritical N<sub>2</sub> Jet      Supercritical N<sub>2</sub> Jet

# Objectives

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## *Overall*

- Determine the mechanisms which control the breakup, transport, mixing, and combustion of sub- and super-critical droplets, jets, and sprays.

## *This Presentation*

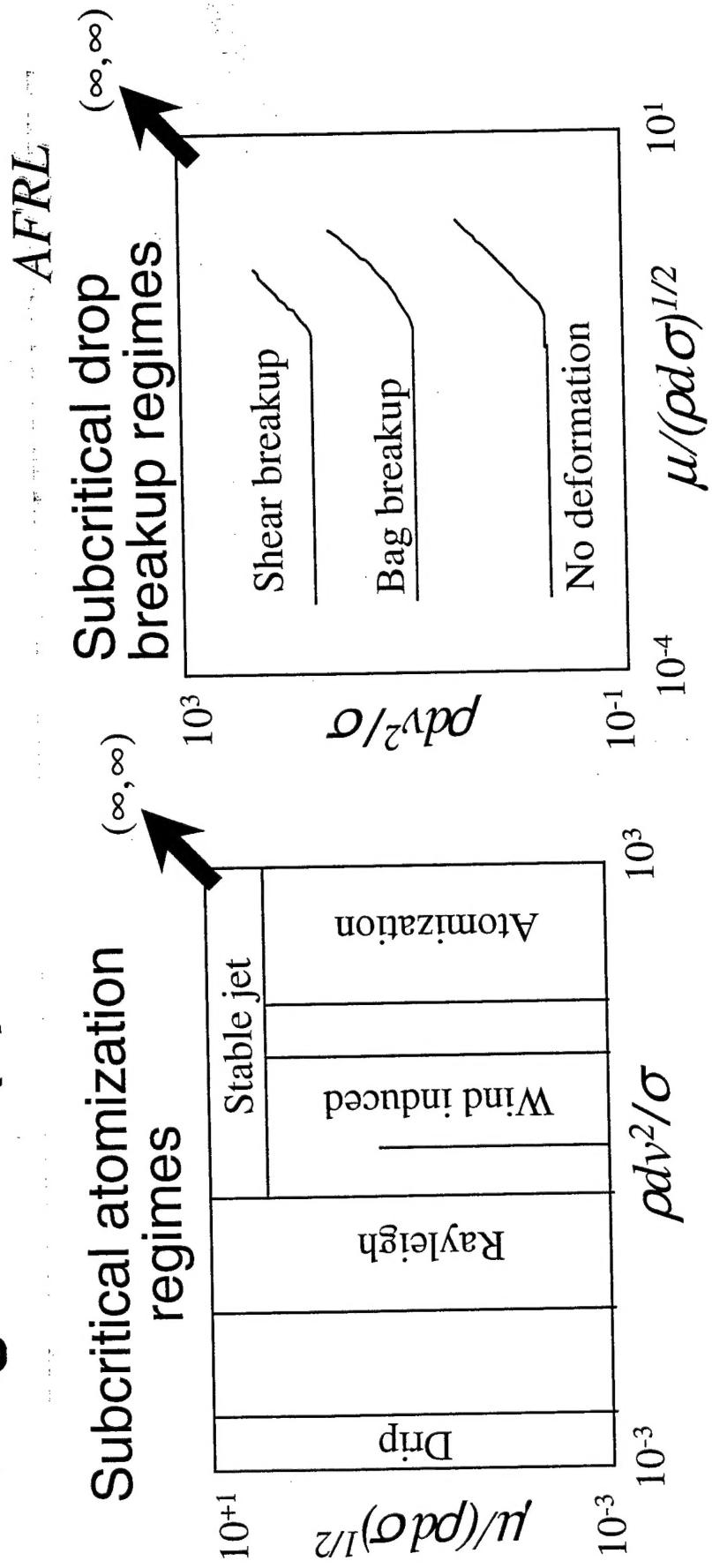
- Determine the structure of subcritical and supercritical cryogenic jets using quantitative Raman imaging.

# Background

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- In engines having chamber pressures exceeding the critical pressure (SSME, Vulcain, etc.), the distinct difference between a “gas” and a “liquid” disappears.
- The resulting flows are influenced by factors not present in conventional sprays:
  - Vanishing surface tension and enthalpy of vaporization.
  - Equivalent gas and liquid phase densities.
  - Strongly enhanced gas / liquid solubility.
  - Liquid-like gas phase diffusivity.
  - Mixing induced critical point variations.
  - Enhanced gas phase unsteadiness.
- *Unknowns contribute to potentially large uncertainties in making design predictions.*

## Background (2)



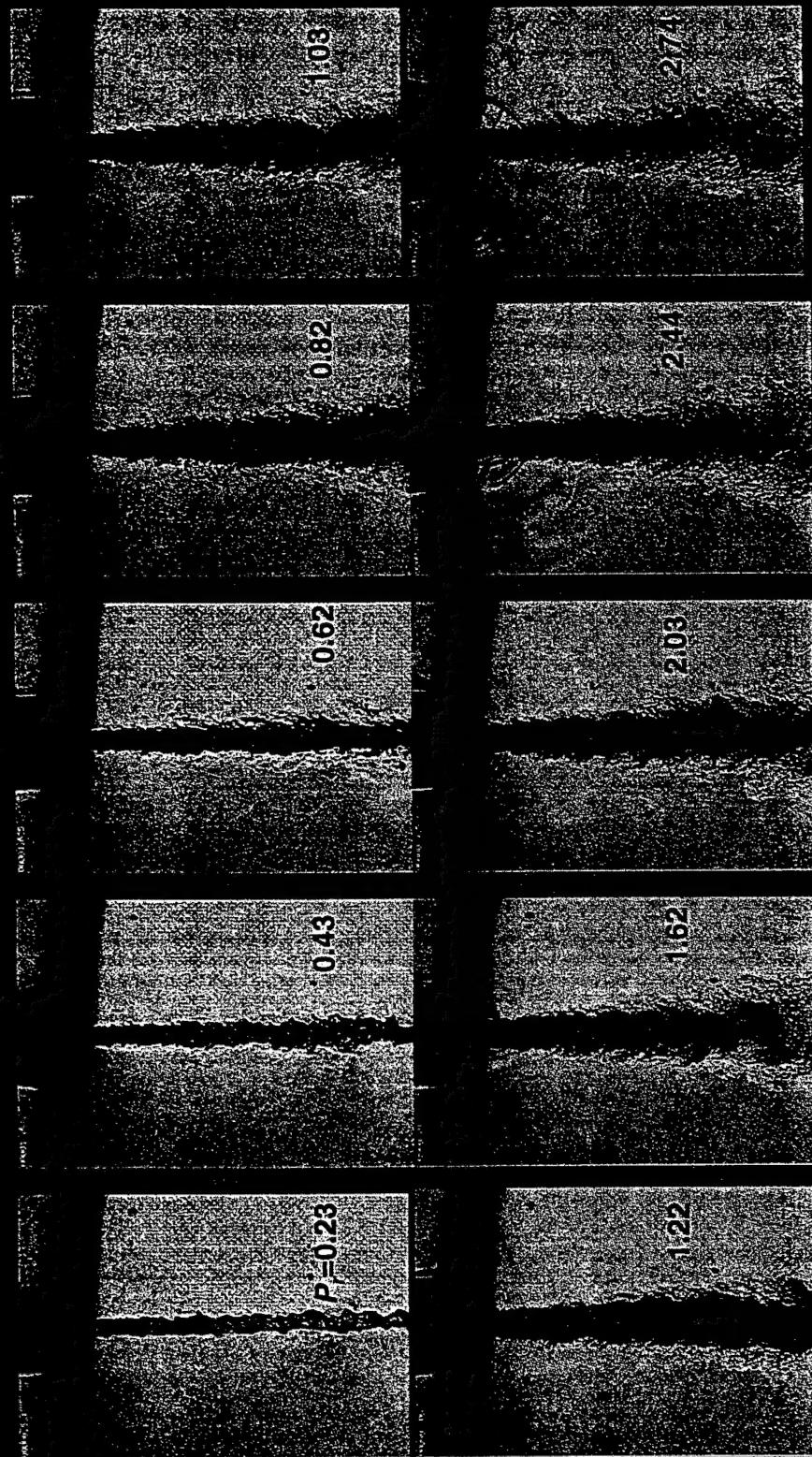
Surface tension  $\sigma$  vanishes at supercritical conditions.  
Conventional atomization and breakup parameters become *infinite*, where no data exists.

*Supercritical* atomization and breakup regimes are largely unknown

# Shadowgraph Results - N<sub>2</sub> into N<sub>2</sub>

P<sub>cr</sub> = 3.39 MPa    T<sub>amb</sub> = 300 K    Re = 25,000- 75,000

T<sub>cr</sub> = 126 K    T<sub>inj</sub> = 99-120 K    V<sub>inj</sub> = 10-15 m/s



# Mixing Layer Structure - N<sub>2</sub> into N<sub>2</sub>

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P<sub>cr</sub> = 3.39 Mpa, T<sub>cr</sub> = 126 K, T<sub>inj</sub> = 128 K, T<sub>amb</sub> = 300 K

Pr=0.91

Pr=1.22



Low Pres.  
Subcritical  
Droplets

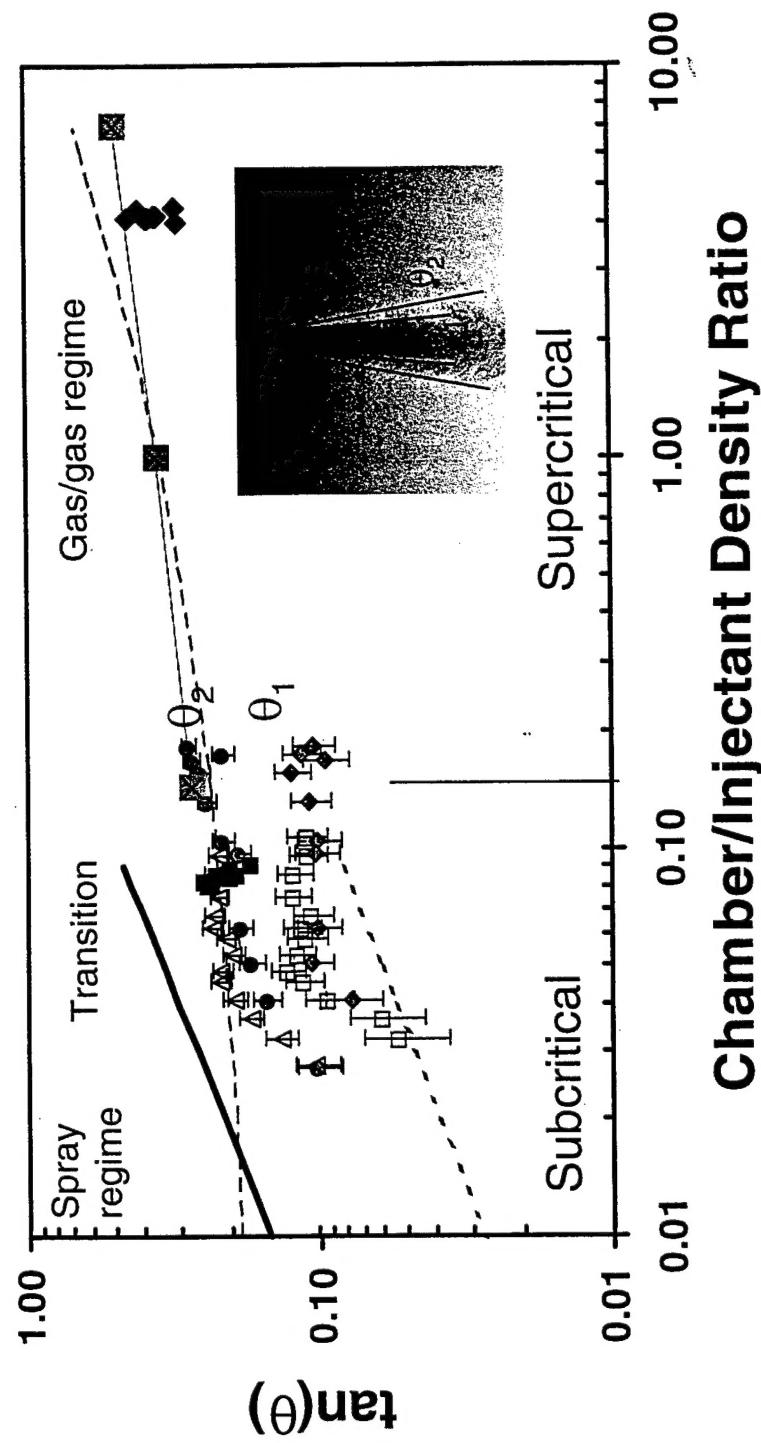
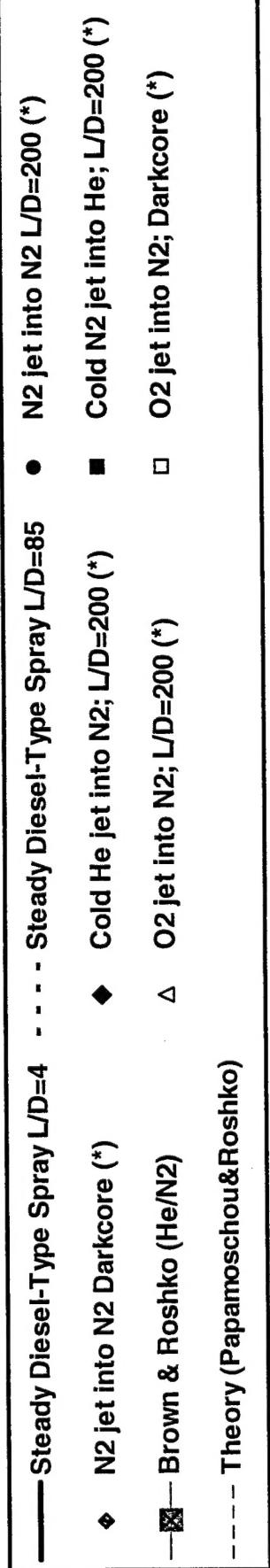
Mod. Pres.  
Supercritical  
Transition

High Pres.  
Supercritical  
Gas layers

# Jet Spreading Angles

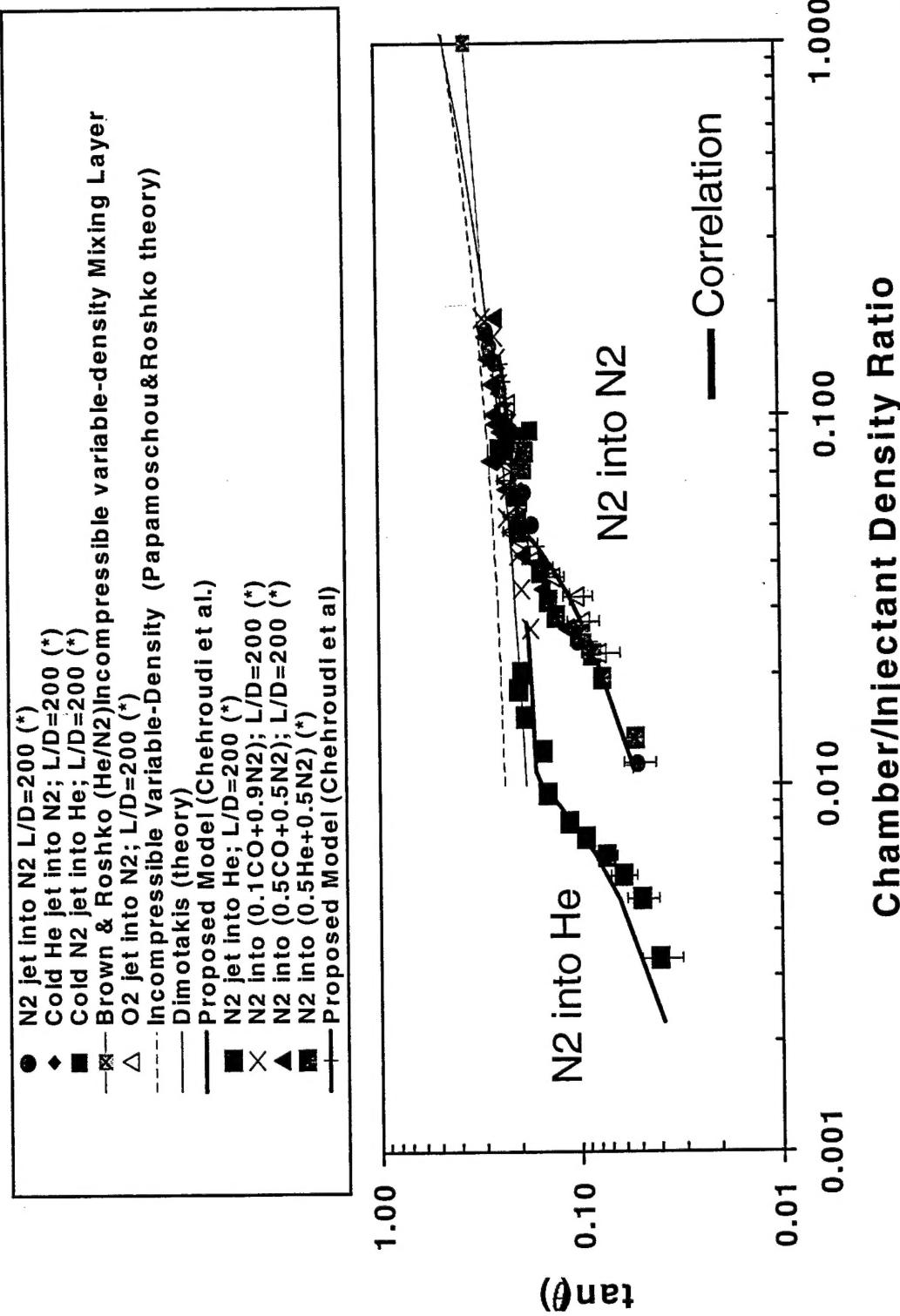
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Chehroudi *et. al.*, AIAA 99-0206, AIAA 99-2489



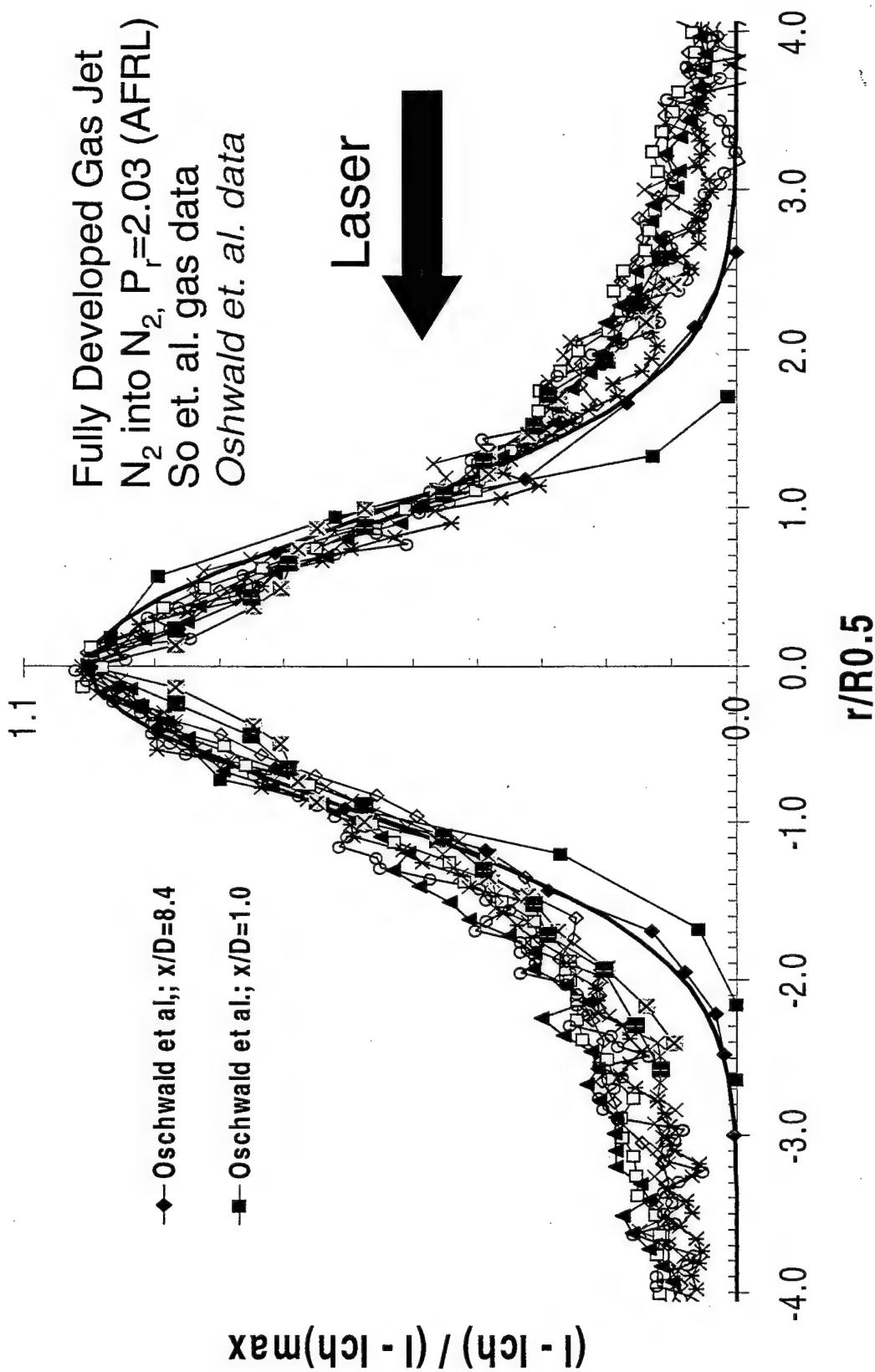
# Empirical Correlation

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# Normalized Intensity Defect Plot: Supercritical Regime (3)

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# Normalized Intensity Defect Plot:

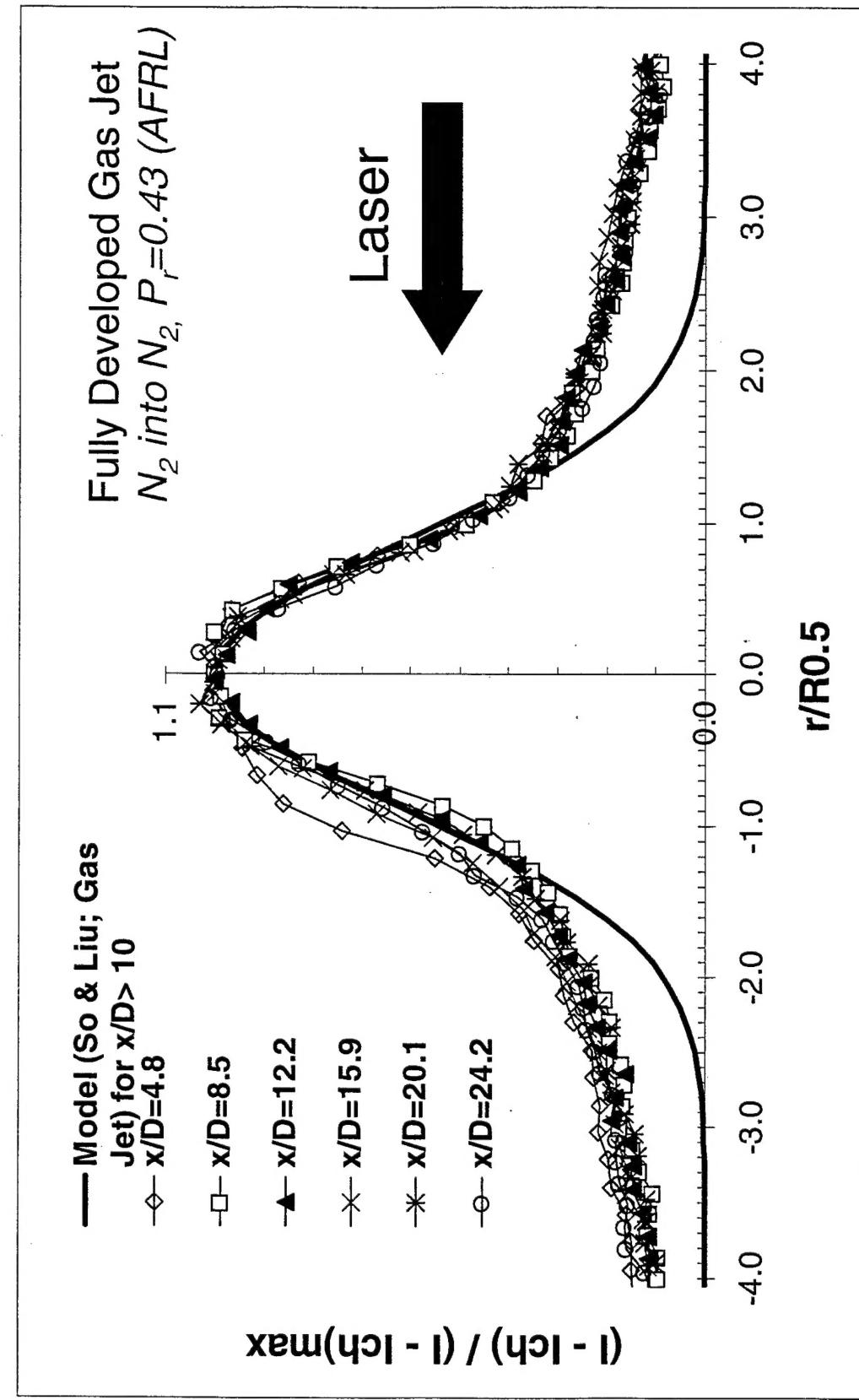
## Supercritical Regime (4)

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	X/D	Pch MPa	Pr	Inj. Temp K	Inj. Vel m/s	Re	Inj/Cham density ratio
Oschwald et al.	1.0	4.0	1.2	140	5.0	115000	3.3
Oschwald et al.	8.4	4.0	1.2	118	5.0	126000	12.5
Chehroudi et al.	4.8 to 24.4	6.9	2.0	95	8.0	35000	7.1
Chehroudi et al.	4.8 to 24.4	1.5	0.4	110	8.0	12000	40.6
So et. al.	5.1	0.1	--	275	11.6	5000	0.6
So et. al.	6.4	0.1	--	275	11.6	5000	0.6

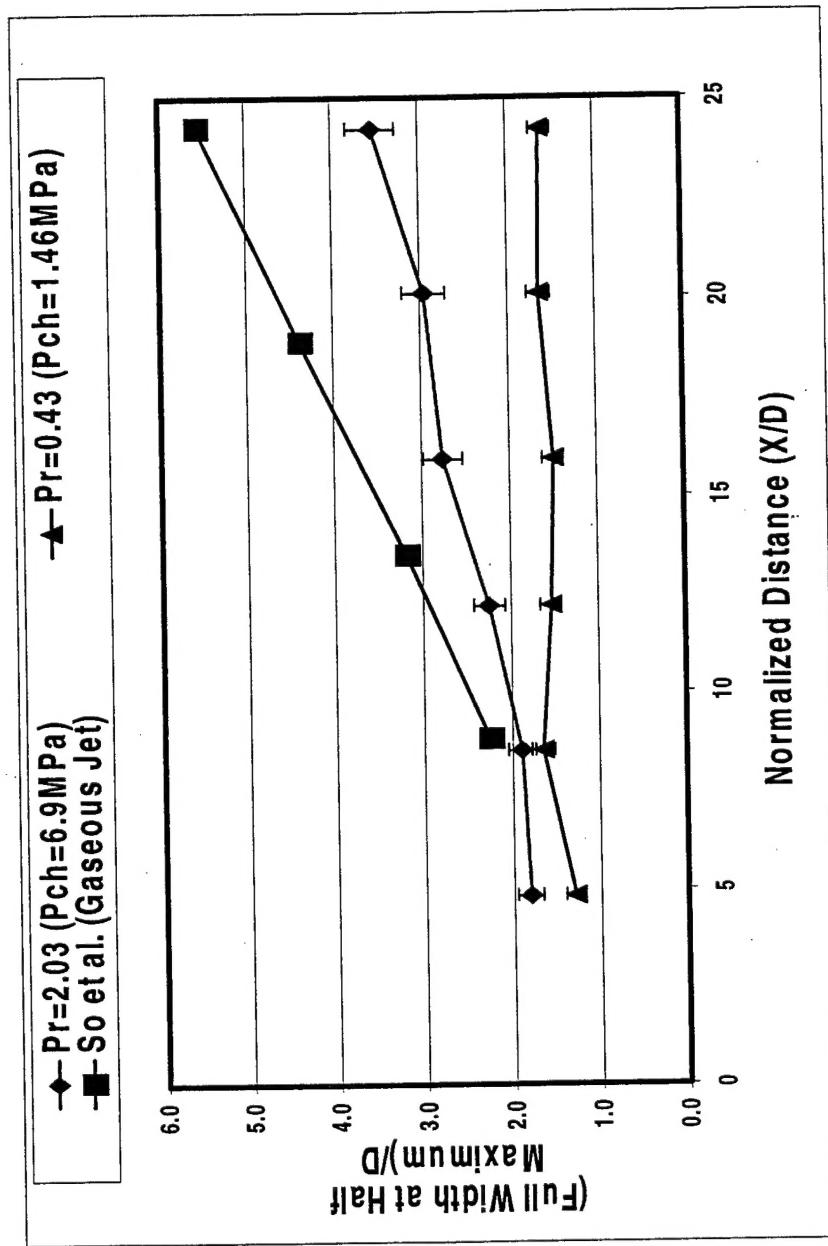
# Normalized Intensity Defect Plot: Subcritical Regime

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# Growth Rates

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# Summary & Conclusions

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- Measurement system integrity has been established by performing Raman measurements of isothermal  $N_2$  at different pressures.
- Measurements were constrained to the near-field in order to maintain large Froude numbers (minimize buoyancy).
- Growth rates measured from Raman profiles measured at 2 x FWHM point agree well with shadowgraph measurements.
  - The equivalency of visual and density growth rates has also been reported in the literature (Brown & Roshko, 1974).
- To within experimental error, the near-field plots appear to reduce to self-similar shapes for both the supercritical and subcritical cases.
  - Not the same profile as for fully developed turbulent gas jets.
- The near-field supercritical profile more closely approaches that of fully developed turbulent gas jets than the near-field subcritical profile.

# Future

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- Complete  $N_2$ -into- $N_2$  analysis.
- Reduce and analyze  $N_2$ -into- $N_2/He$  data.